

Fig 4 1974 double layer test 1981 double layer solar 1999 multi layer Breitling world flight

Parameters for a ten-layer Titan balloon and corresponding heat loss			
Altitude 10,000 m	Ambient 84 °K	Density 3.3 kg/m3	6 m internal diameter
Outer fabric 17 gsm	9 inner layers 10 gsm	Conductivity ratio 2	7 m external diameter
Payload 100 kg	ASRG 22 kg	Envelope 12.3 kg	Lift temperature 139 °K
Heat Loss 210 Watts. An ASGR provides 360 Watts [thermal], a substantial surplus			

Industry consortium. Improved Titan design. Multi-layered: plenty of precedents.

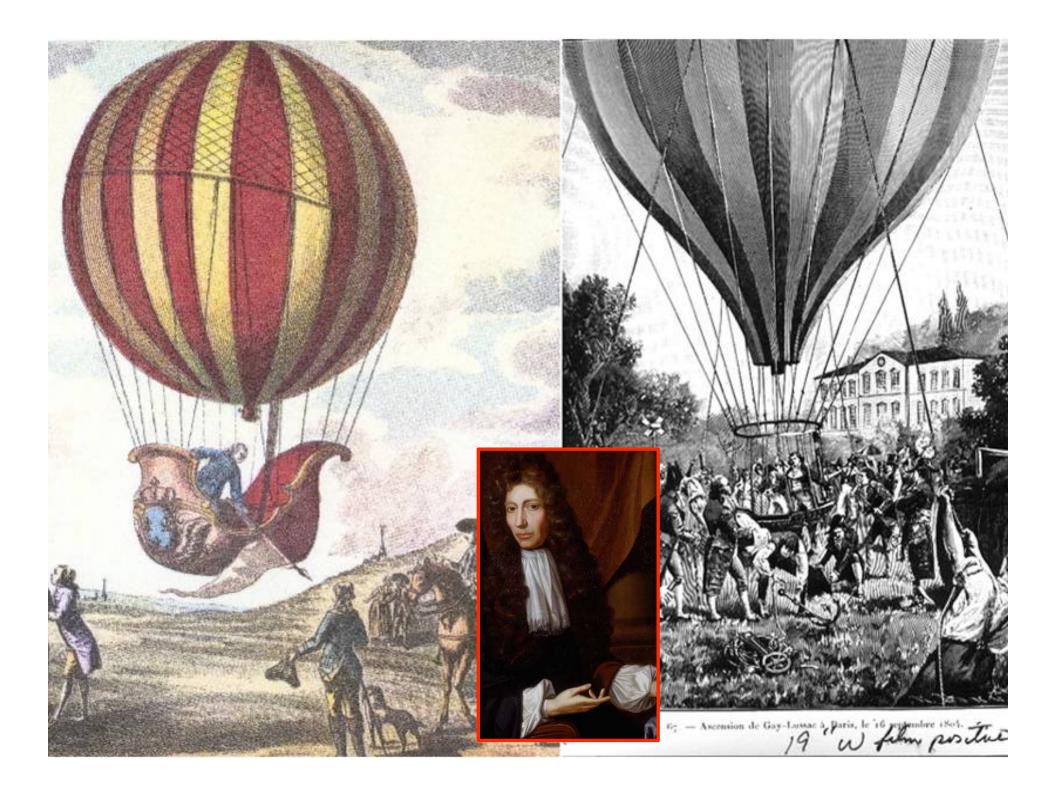
This is a work in progress.

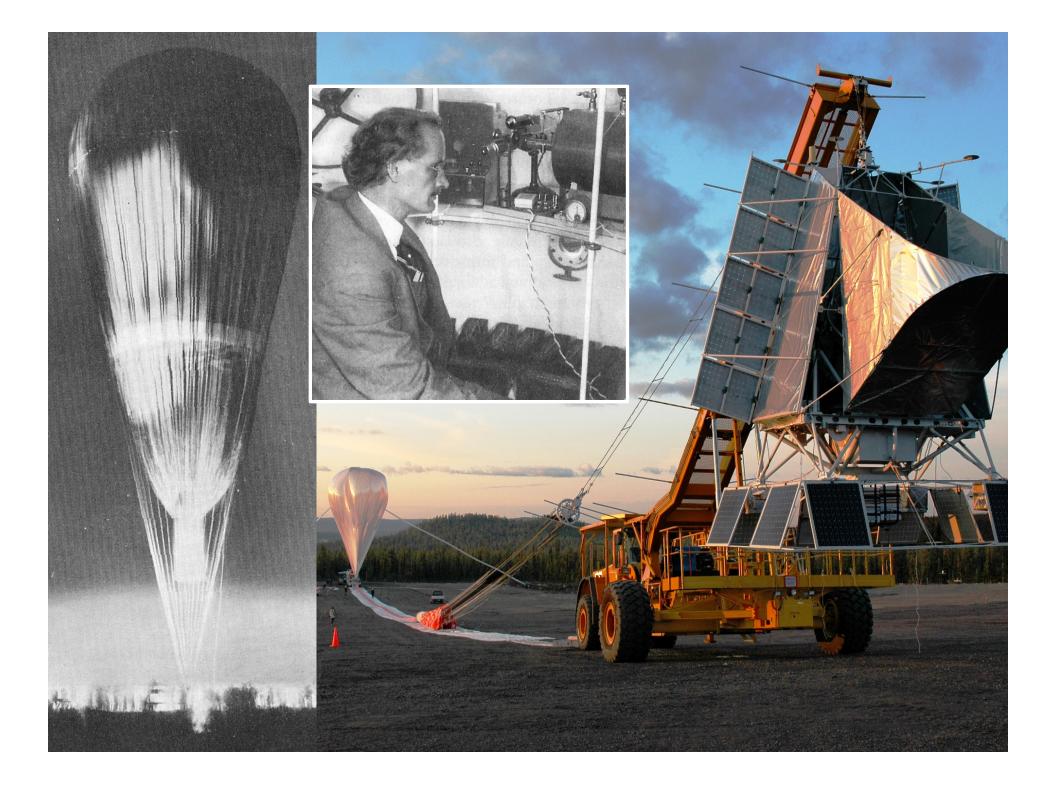
But this is a workshop.

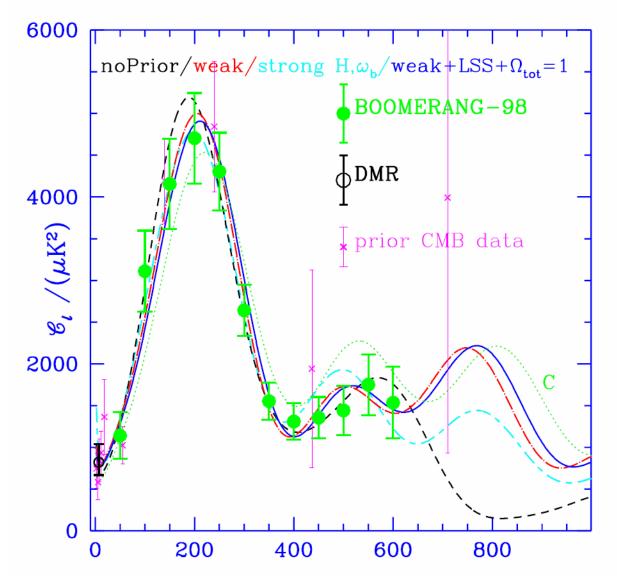
Full paper in September at AIAA.

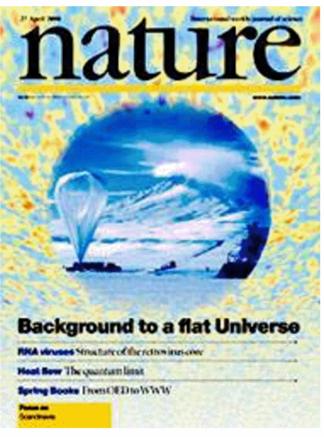
One ASRG. But Decadal recommends even JEO should use ASRG.

Design to be testable.









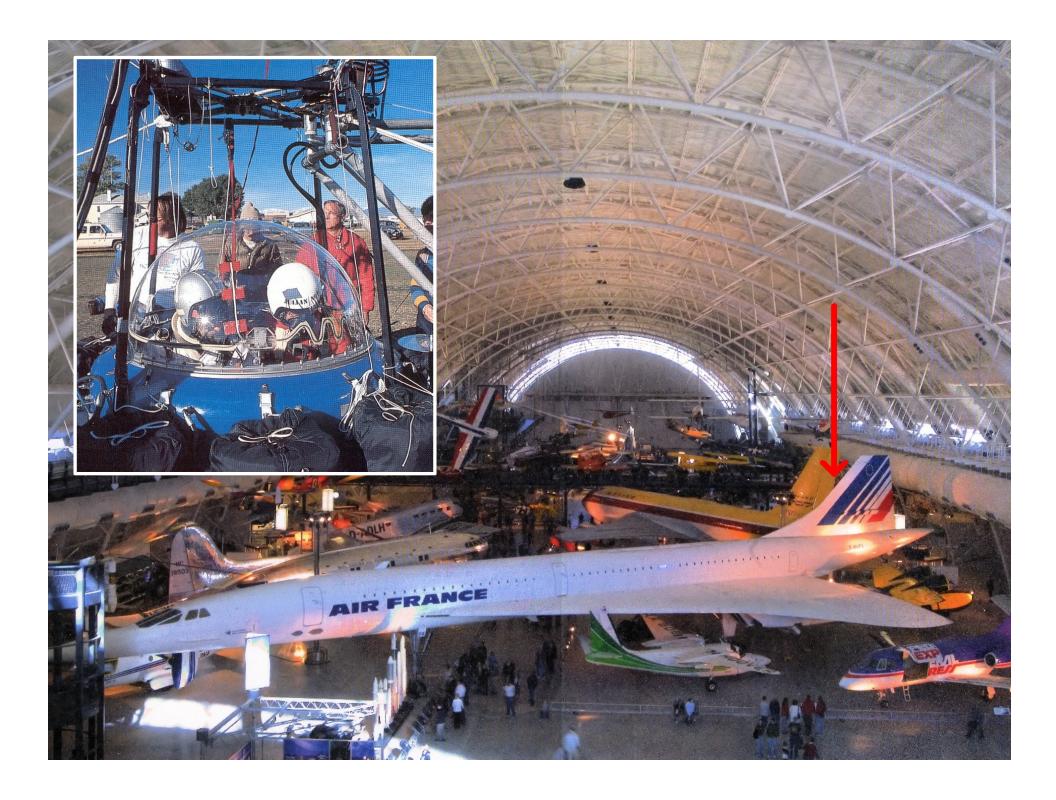


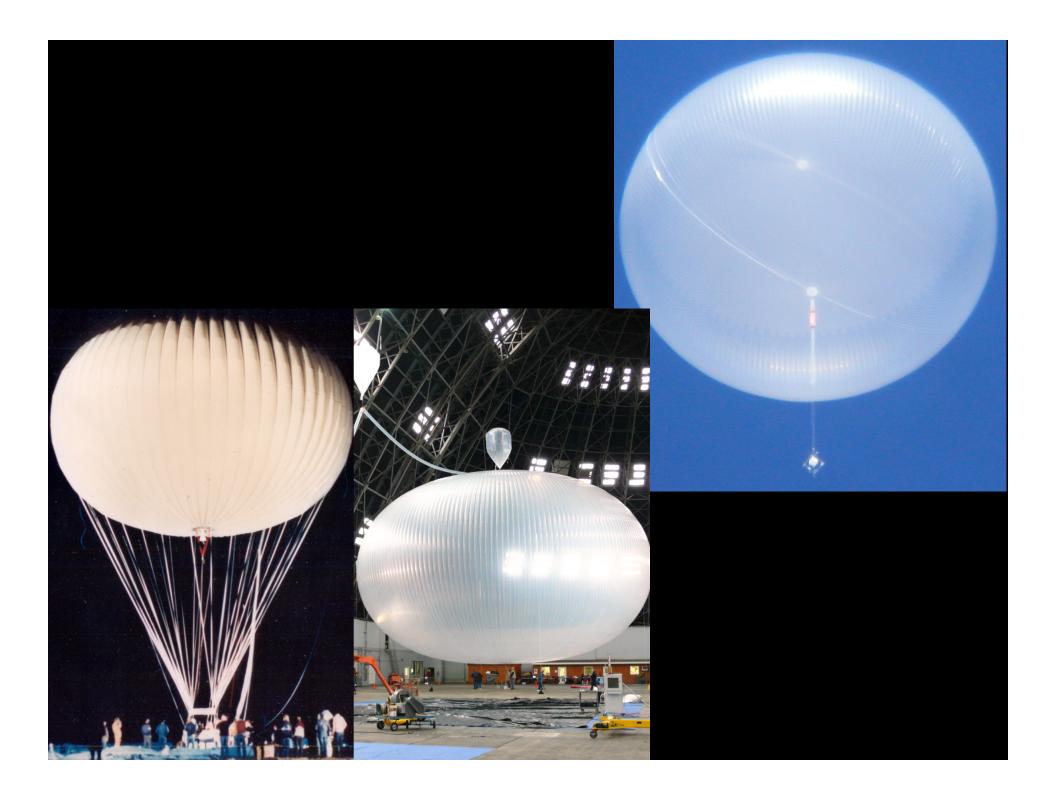


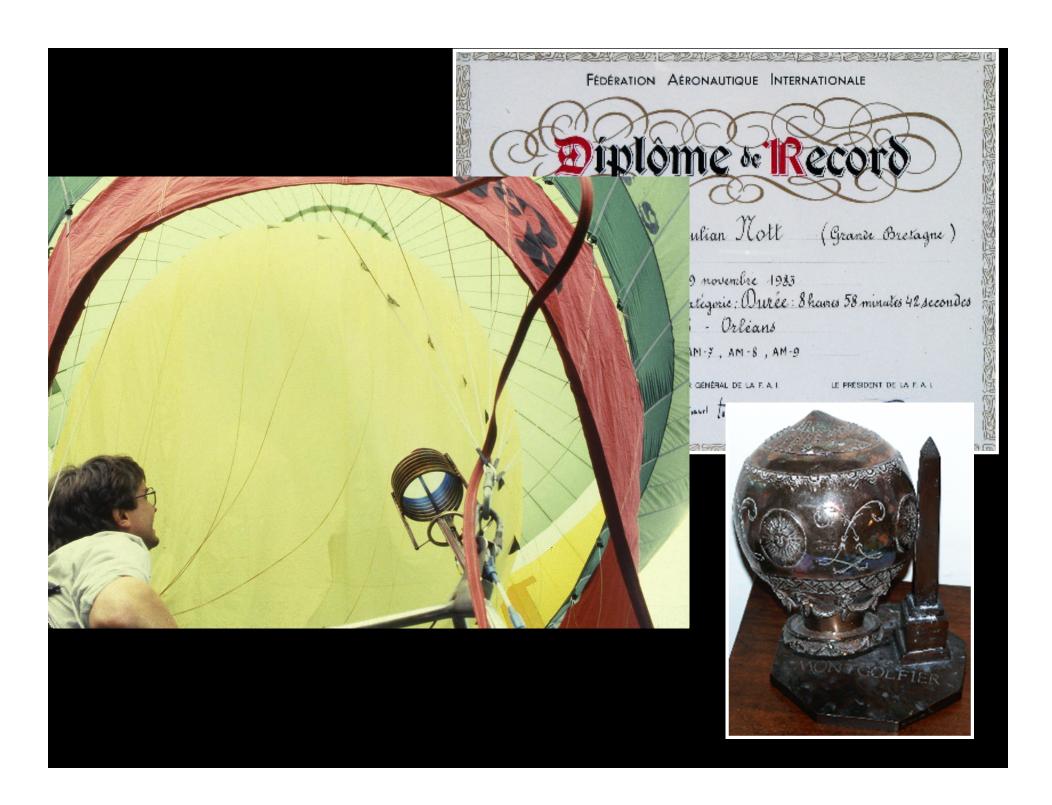


"A ship in harbor is safe, but that is not what ships are built for."

John A. Shedd







Space scientists worry that balloons are novel and hence risky.

They should not be thought of as novel spacecraft, but balloons that simply arrive via space.

Balloons are extremely familiar. 50 million weather balloons. At least 4 million hot air balloon flights. Tens of thousands of flights by hydrogen and helium balloons.

Conditions at potential Solar System destinations are well within terrestrial experience.

Titan's conditions are highly favorable and familiar for balloons.



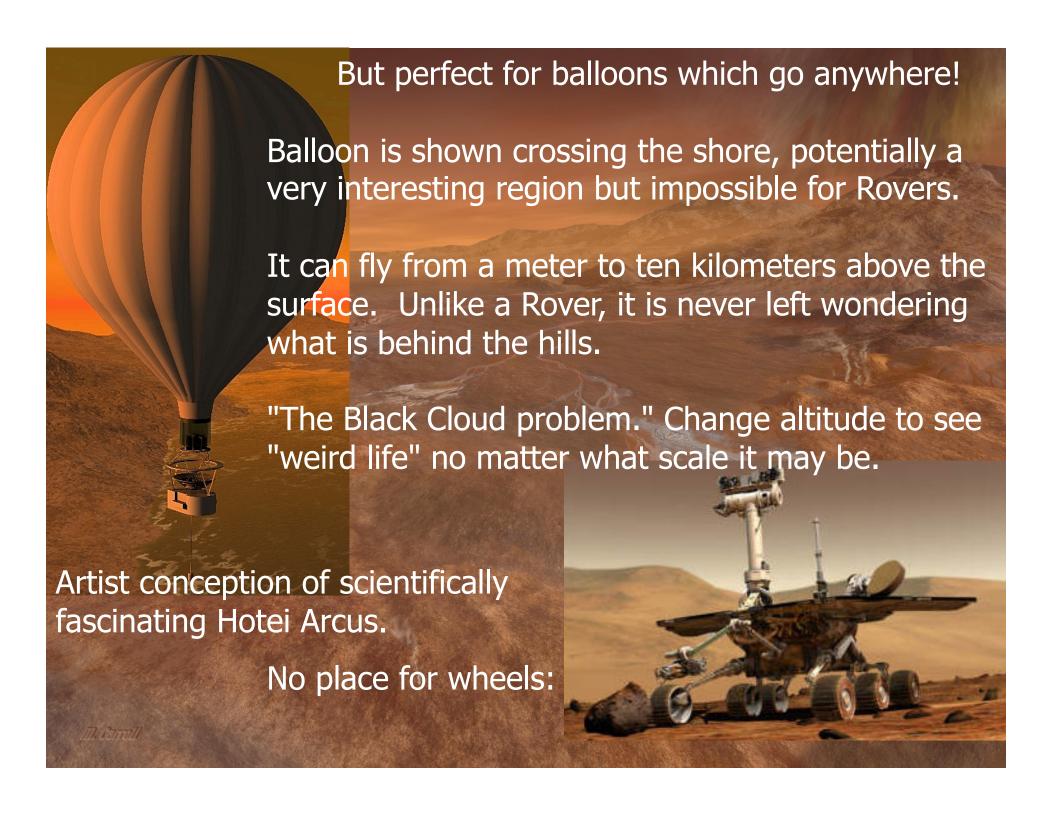
Titan conditions are a 1,000 times better for balloons than Earth conditions.

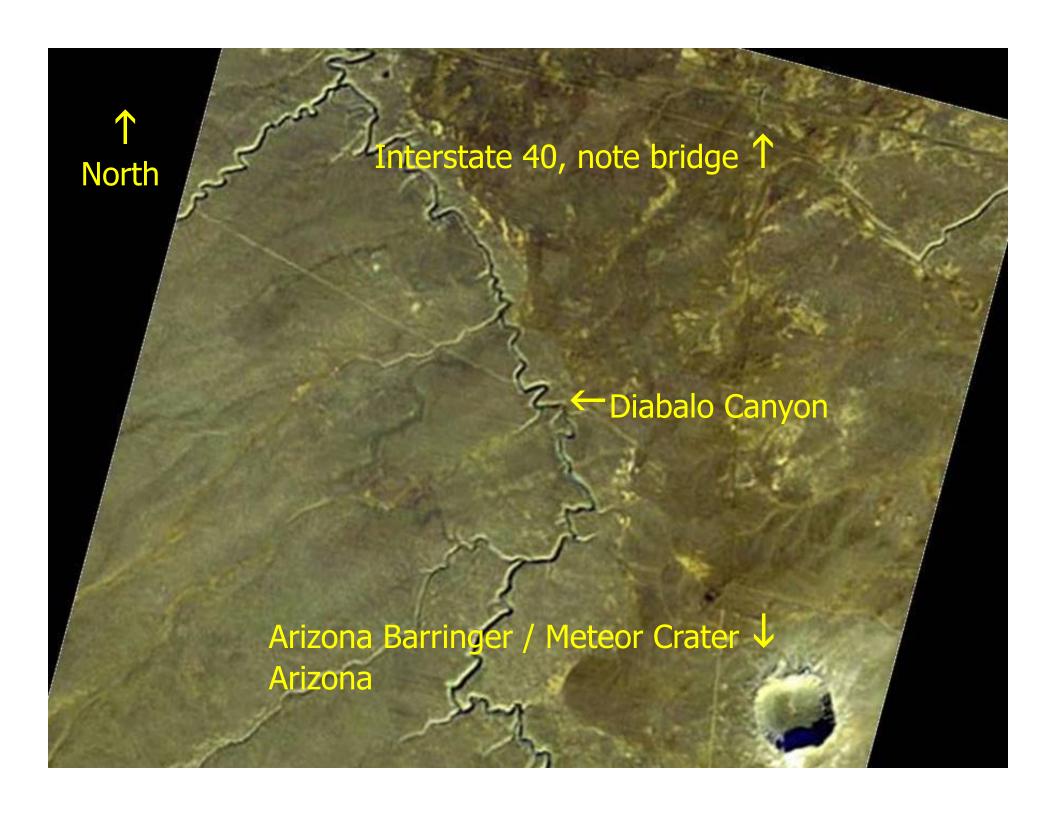
cold: diffusion many orders of magnitude less.



15 times greater for a given temperature rise.

VERY LOW TEMPERATURE
VERY HIGH DENSITY
LOW GRAVITY







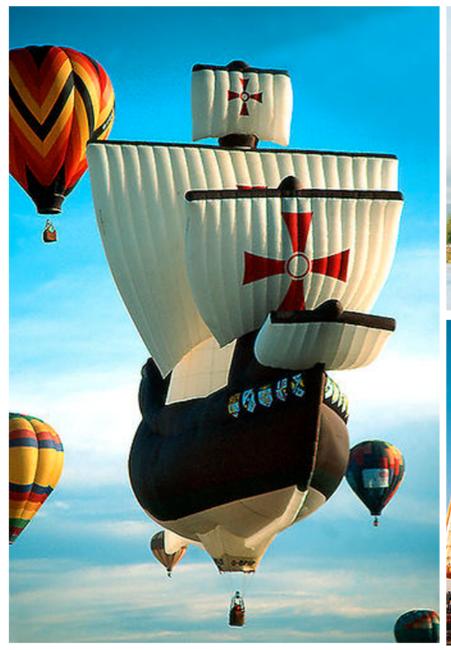








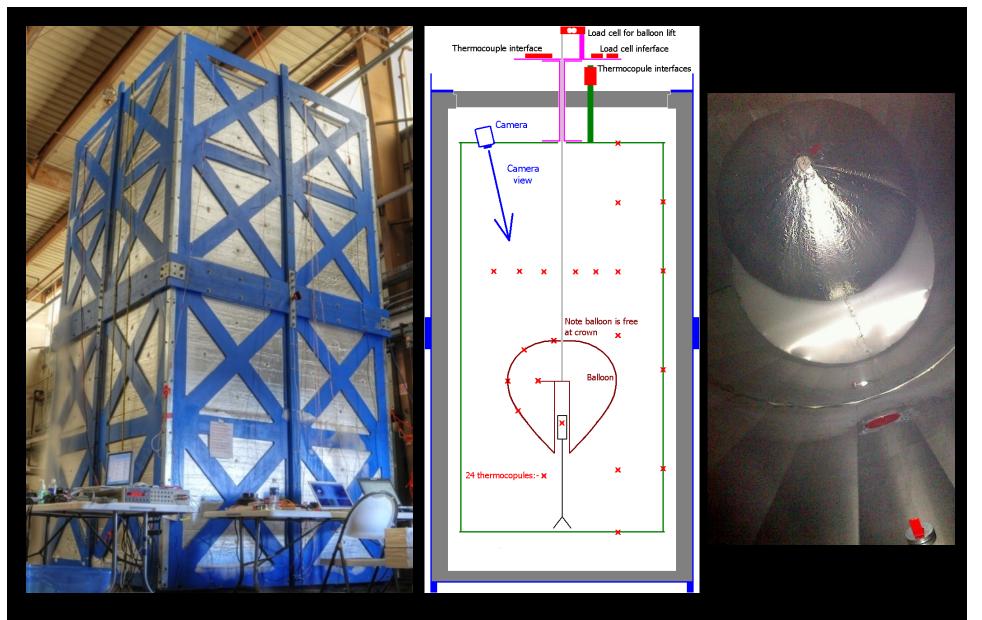






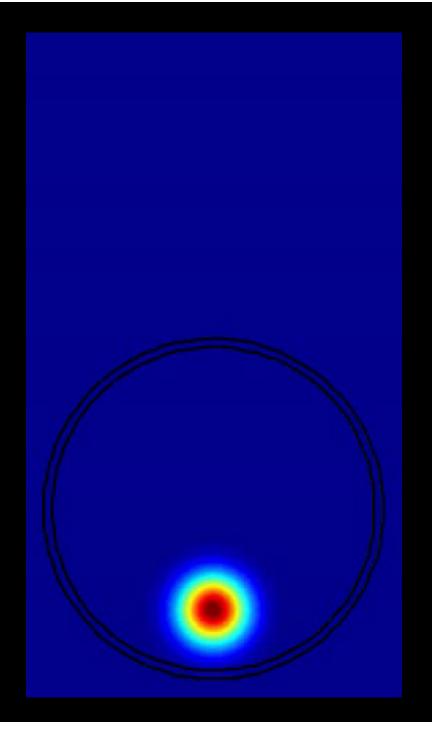


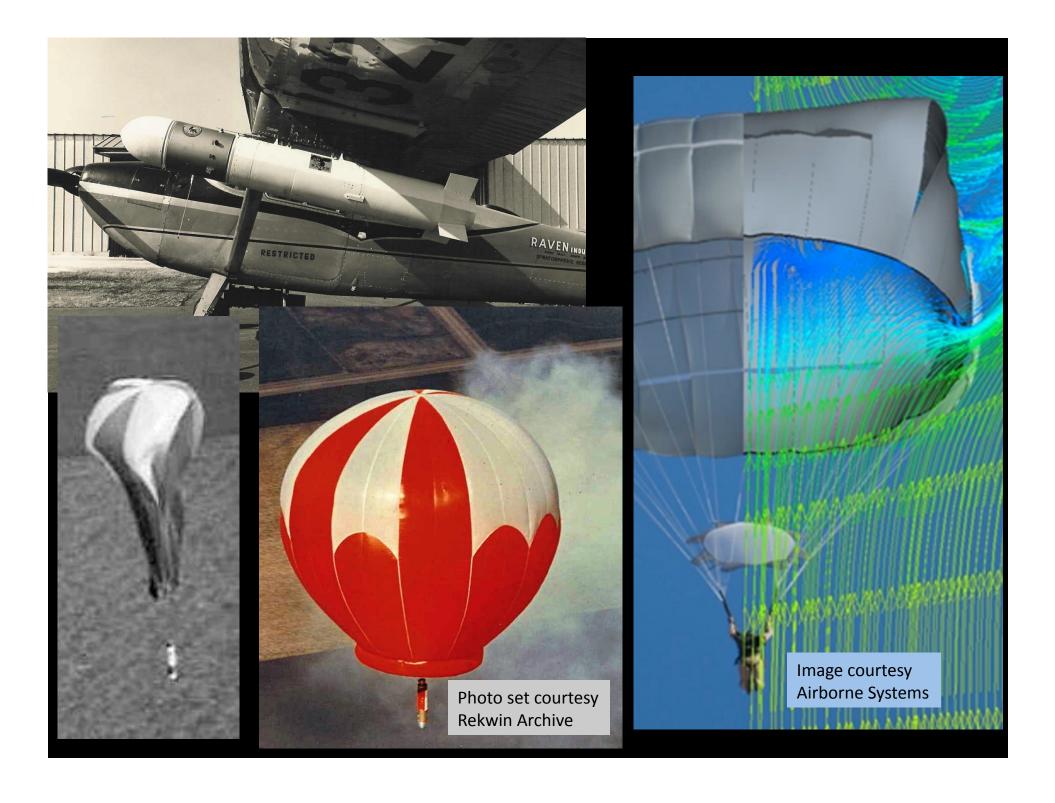




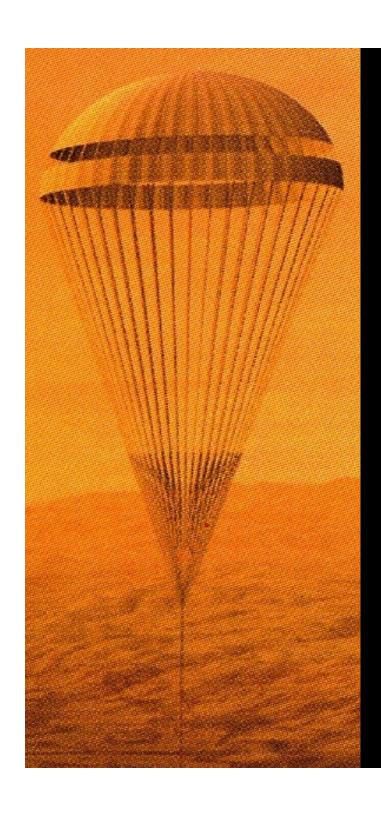
Within the limits of the experiment, the qualitative behavior is the same from this tiny balloon at cryogenic temperatures to a balloon 70,000 times larger.

Computational Fluid
Dynamics [CFD] modeling
of Titan balloon convection
by Professor Tim Colonius,
Caltech.









Rob Sinclair Chief Engineer, Airborne Systems Space Group

Former test jumper
Designed Huygens Parachute
Design of CEV Parachute

The Source for FSI

FSI = Fluid, Structure Interaction

WEATHER CONDITIONS

With any aircraft design and especially lighter than air craft, the first thing you ask is what are the weather conditions?

"Sail on the Central Park reservoir or around Cape Horn?"

Can't decide most basic things, even fabric strength.

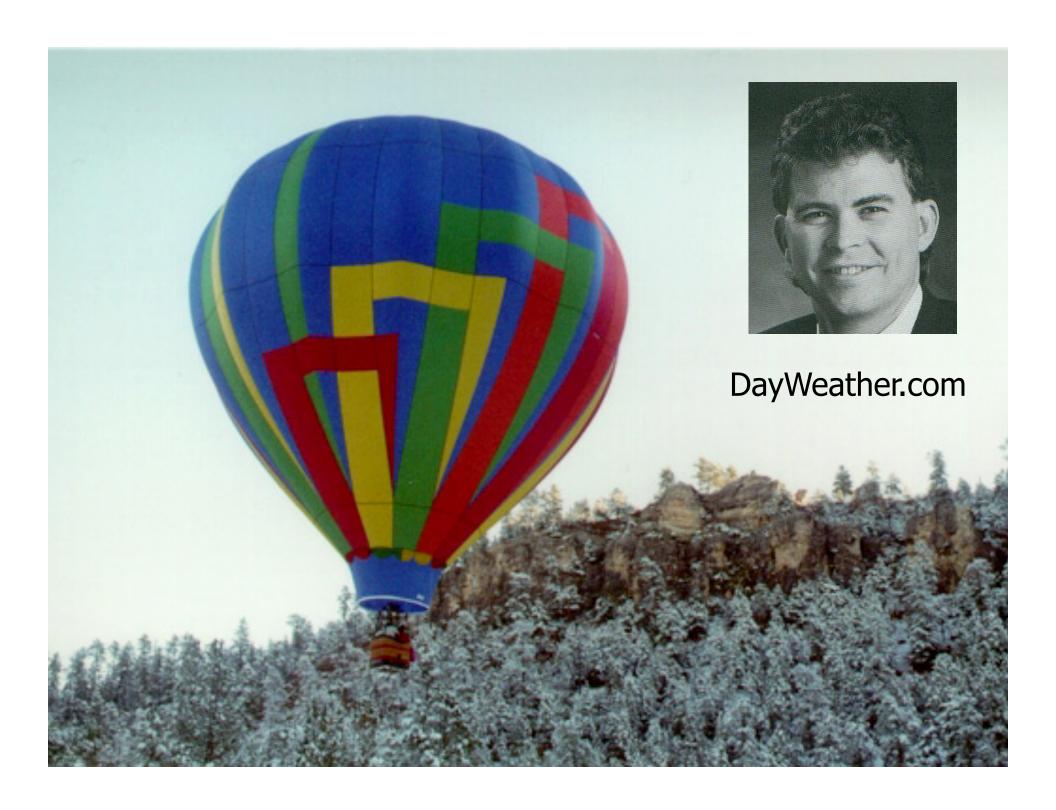
Start at the safest zone for the season.

Steering objectives.

Local and/or short-term phenomena especially convective clouds. Atmosphere temperature variation of short time and distance. Lapse rates.

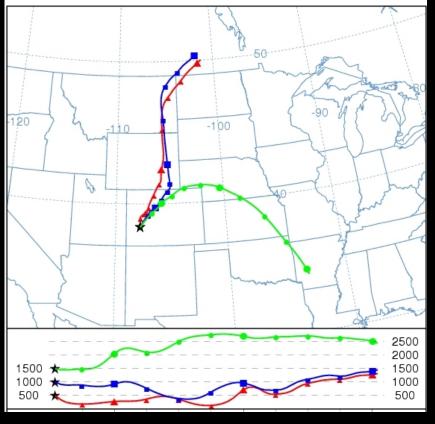
STRATEGY FOR DEALING WITH UNCERTAINTY – CRUCIAL IN MANNED FLIGHTS

State of the art ... which is evolving



STEERING

NOAA HYSPLIT MODEL
Forward trajectories starting at 13 UTC 09 Oct 07
06 UTC 09 Oct GFSG Forecast Initialization



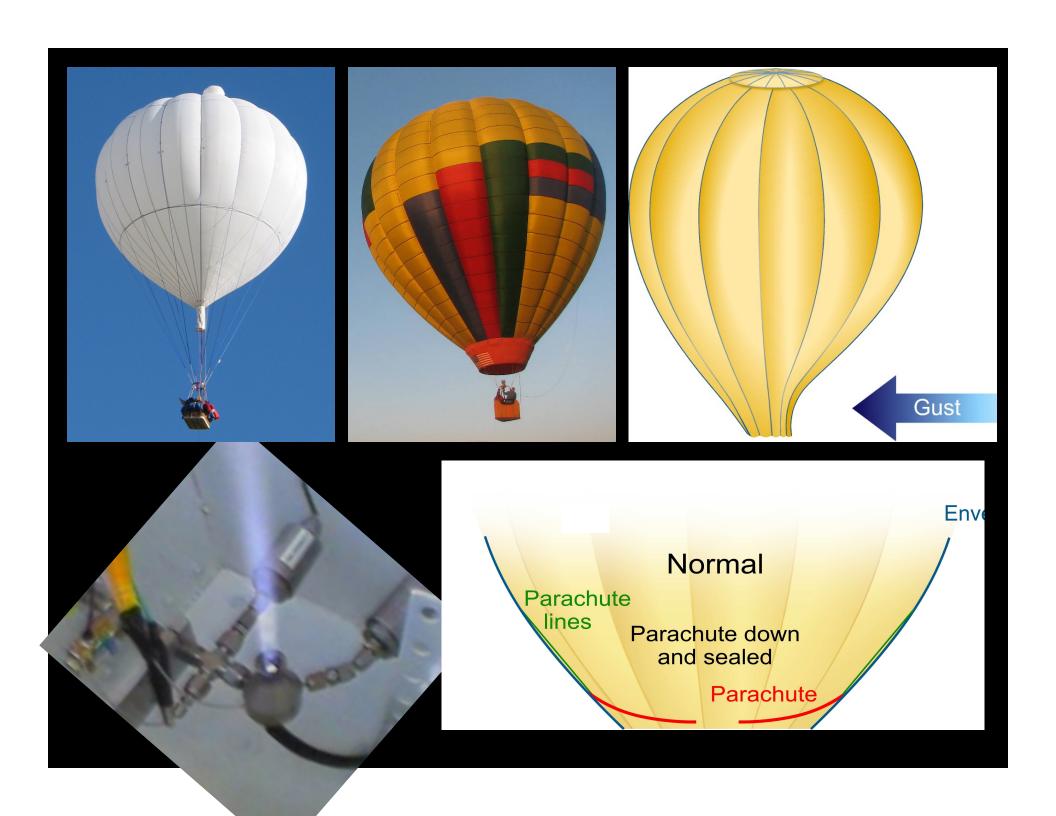


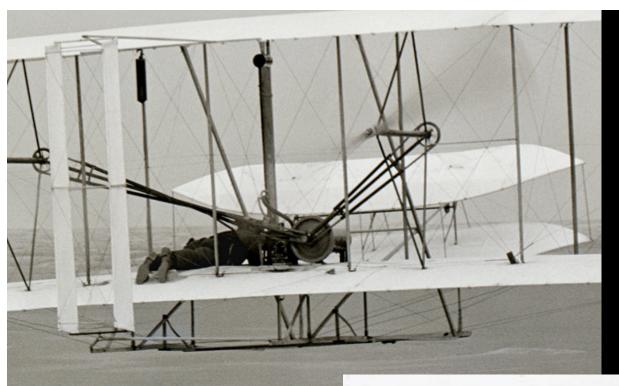
Continental to global scale steering with wind models.

Propellers a minor penalty and a major benefit

"Autonomous" steering over 10 kilometers

Source: Julian Nott / www.NOTT.com





1903

1908: Just 5 years.





1908



2008 - full 100 years ge.o.graphy ocean.o.graphy

titan.o.graphy

titanography.com up by July 4 Weekend

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